

Retraction

Retracted: Assessing the Feasibility of Cogeneration Retrofit for Heating and Electricity Demands in Marine Diesel Engines

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This article has been retracted by Hindawi, as publisher, following an investigation undertaken by the publisher [1]. This investigation has uncovered evidence of systematic manipulation of the publication and peer-review process. We cannot, therefore, vouch for the reliability or integrity of this article.

Please note that this notice is intended solely to alert readers that the peer-review process of this article has been compromised.

Wiley and Hindawi regret that the usual quality checks did not identify these issues before publication and have since put additional measures in place to safeguard research integrity.

We wish to credit our Research Integrity and Research Publishing teams and anonymous and named external researchers and research integrity experts for contributing to this investigation.

The corresponding author, as the representative of all authors, has been given the opportunity to register their agreement or disagreement to this retraction. We have kept a record of any response received.

References

- [1] X. Liu, H. Yang, Z. Feng, G. Hu, and Z. Zeng, "Assessing the Feasibility of Cogeneration Retrofit for Heating and Electricity Demands in Marine Diesel Engines," *International Transactions on Electrical Energy Systems*, vol. 2023, Article ID 3280579, 10 pages, 2023.

Research Article

Assessing the Feasibility of Cogeneration Retrofit for Heating and Electricity Demands in Marine Diesel Engines

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In the marine engineering industry, turbocharged diesel engines are often used to generate electricity, and hot oil can be extracted after generating electricity. However, marine diesel engine heat recovery can be distinguished from gas heat recovery for turbocharged and nonturbocharged diesel engine systems. The ideal air model Brayton cycle is used to evaluate the feasibility of turbocharged/nonturbocharged cogeneration retrofits in turbocharged diesel engine systems, and the paper is designed to evaluate the effect of pressure and temperature and cooling ratio of exergy efficiency. The results show that the performance of turbocharged and nonturbocharged work increases with increasing pressure ratio until it reaches a maximum value and decreases with increasing pressure ratio at a constant temperature. If an electric generator is selected first, the heat recovery after the turbocharger can be used to improve the contract for heating and electricity needs. While the exergy efficiency is selected priorly, cogeneration retrofit for heating and electricity demands can be used for heat recovery without the turbocharger system.

1. Introduction

Many turbocharger diesel engines are provided the power for the need of marine engineering (including FPSO (the floating production storage and offloading facility) and oil extraction platform). The extracted oil and other heat users of marine engineering should be offered huge quantity of heat. Then, the cogeneration retrofit for heating and electricity of diesel engines with and without a turbocharger is the way to solve the heat loss problem of released flue gas.

In order to use the heat loss of turbocharged and nonturbocharged diesel engines, many researchers focus on the study of the system. Many researchers focus on the ORC process to utilize engine power loss. Ma et al. designed an ORC waste heat recovery system that uses a water jacket as a preheat source and exhaust gas as an evaporative heat source to treat waste heat from special internal combustion engines. The system can increase the efficiency of the diesel engine by 14.23% when R141b is used as the working fluid. Yang et al. [2], Wang et al. [3], and Song et al. [4] have

conducted research on the dual loop ORC (DORC) of internal combustion engine waste heat recovery. The research results show that DORC can effectively improve the efficiency of internal combustion engines. It improves its economy. Mosaffa et al. [5] compared to the organic Rankine cycle, the gas recycling cycle, and the organic Rankine regeneration cycle and found that the regeneration cycle can achieve the highest efficiency, and the recycled gas achieves the highest efficiency of 39.93% whose prices increased only by 5.2%. Bombarda et al. [6] analyzed the effects of system vapor pressure, vapor temperature, and expansion on the Kalina cycle and ORC performance in a typical diesel engine operation. The results of the study show that the ORC system is suitable for heat recovery in internal combustion engines. Mago [7] and Vaja [8] discussed the effect of space reheating on ORC performance from several perspectives. The research results show that the thermal process of the recycling center can improve the thermal efficiency and energy output of the ORC system. It can also reduce the inevitable loss of the system. Menel et al. [9] proposed an ORC system using

a store-retrieval technique. The system can recover the latent heat of the superheated working water through the heat storage process. The thermal efficiency of the system is higher than the traditional ORC system and can increase by 2.25%. Yang et al. designed a dual loop organic ordered cycle (ORC) system for waste recovery in 6-cylinder diesel engines, and it has been shown that the maximum energy saving of dual-circuit ORC system efficiency (WHRE) can be conducted at 5.4%.

While the ORC system is too complex for offshore oil extraction facilities, some researchers use the turbocharger to reduce the temperature of the flue gas of diesel engines. Hopman has conducted research on diesel engine turbocharged power generation technology [11, 12]. The research results show that the use of turbocharged power generation devices can increase the fuel economy of diesel engines by 5%. Patterson et al. [13] used the power generation compound turbine technology combined with the high-speed motor to recover the energy of the low-pressure stage turbine of the two-stage supercharged diesel engine. For MAN Company, it is proposed to connect the turbocharger and the electric turbine in parallel, which reduces the gas flow from the turbocharger so that part of the exhaust gas, after leaving the cylinder of the diesel engine, goes directly to the turbocharger to work [14]. This method of power turbine can increase the output of the engine by 3–5% according to SMCR. Wärtsilä [15] created a similar electronic device that uses fuel separation before the turbocharger so that the rest of the exhaust gas, after leaving the engine cylinder, goes directly to the electrical work. Taking Wärtsilä 14RT-flex96C (maximum sustainable rated power 80080 kW) as an example, the research results show that, without reducing the engine output power, 10% of the engine exhaust gas will directly enter the power turbine without passing through the turbocharger. The power turbine can emit about 2500 kW of power under 100% CMCR working conditions, accounting for 3% of the engine output, which effectively increases the efficiency of the ship's engine. Caterpillar and John Deere proposed the use of electric compound turbines (ETC) to replace turbochargers and power turbines. Studies have shown that fuel consumption can be predicted to be reduced by 5–10% after the use of ETC devices [16, 17].

From the practical operation, the flue gas temperature is just depressed above 300°C and the remaining heat can be used with heat recovery boilers. Jayakumar et al. [18], to analyze the thermal performance, used the ideal air model Brayton cycle to remove the fan of the boiler engine to compensate for the waste due to the limited space of the coast limited compared to the design.

2. System Description and Assumptions

As shown in Figure 1, the oil is used as a fuel for diesel engines to generate power, and the waste heat is used by boilers and sent to consumers for heating in marine

engineering. First, the gas enters the turbocharger of the diesel engine to increase the air pressure. The flue gas is then discharged into the thermal boiler, and the heat obtained is used to heat the consumer. Physical description includes working fluid: working water is the best fuel; flue mass flow is the same as the air entering the furnace, and the specific heat of the working water is constant.

Brayton air cycle was used in the study of debris removal from turbo diesel engine to evaluate the feasibility of marine engineering contract, according to the work of Liu et al. [19]. Figure 2 shows the corresponding T-S diagram of the Brayton cycle with and without turbocharging [20].

2.1. Mathematical Modeling. Table 1 introduces and expounds on analyzing the devised plant from various standpoints [21].

The various standpoints are determined in Table 2.

The output power W (kW) of the system is written as

$$W = \eta_G q_m c_p [(T_3 - T_2) - (T_4 - T_1)], \quad (1)$$

where q_m and C_p are, respectively, flow mass of flue gas and the specific heat of flue gas.

The exergy of turbocharger E_T (kW) can be written as

$$\begin{aligned} E_T &= q_m C_p \eta_T (T_4 - T_6) \\ &= \frac{q_m C_p \eta_T \beta T_1 (\theta_C - 1)}{\eta_C}, \end{aligned} \quad (2)$$

where the efficiency of turbocharger η_T is 0.5.

Then, (1) can be calculated as

$$W = \eta_G q_m c_p T_1 \left[\alpha \eta_D \left(1 - \frac{1}{\theta_C \rho_{Com} \rho_R} \right) - \frac{\theta_C - 1}{\eta_C} \right], \quad (3)$$

where η_G is the efficiency of the electric generator.

The exergy of heat recovery after turbocharger is E_Q , which is given as follows:

$$\begin{aligned} E_Q &= \int_{T_5}^{T_6} q_m C_p \left(1 - \frac{T}{T_1} \right) dT \\ &= q_m C_p (T_6 - T_5) - q_m C_p T_1 \ln \left(\frac{T_6}{T_5} \right). \end{aligned} \quad (4)$$

The fuel energy (combustion processes) exergy E (kW) of the system can be written as

$$\begin{aligned} E &= q_m C_p (T_3 - T_2) \\ &= q_m C_p T_1 \left(\alpha - 1 - \frac{\theta_C - 1}{\eta_C} \right). \end{aligned} \quad (5)$$

2.2. Mathematical Model with the Turbocharger System without considering Heat Recovery. The relationship of exergy efficiency ε with the compressor pressure ratio θ_C and temperature ratio parameter α is

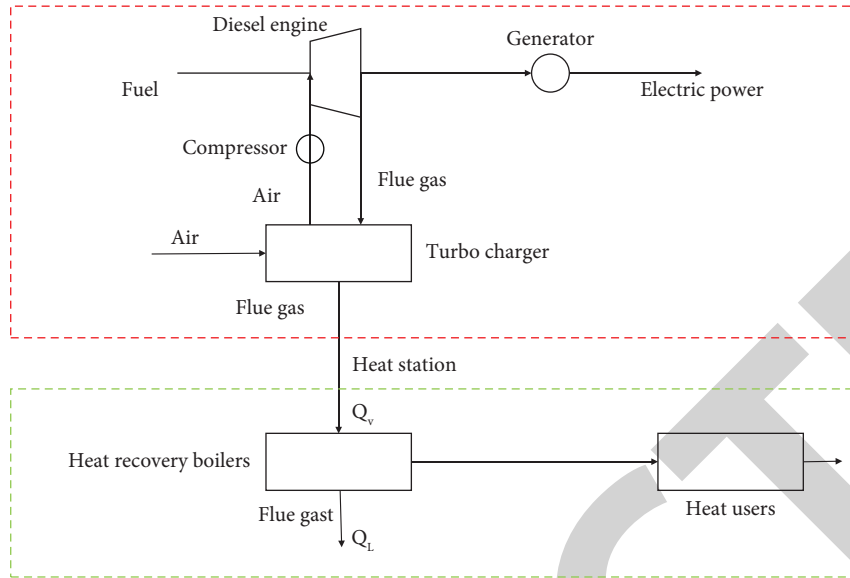


FIGURE 1: Cogeneration retrofit with a turbocharger and heat recovery boilers.

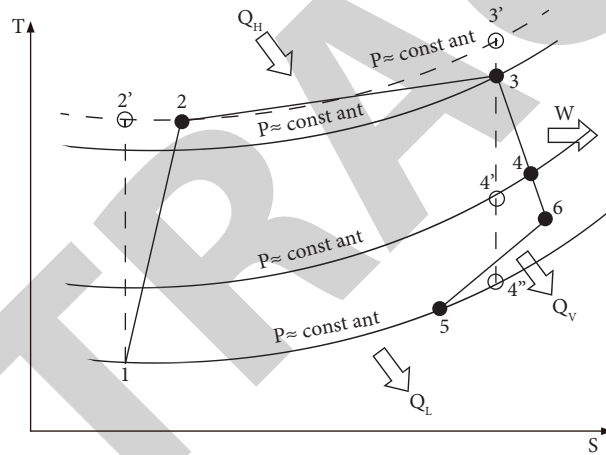


FIGURE 2: System description for turbocharger diesel engines.

TABLE 1: Thermodynamic parameters of the Brayton system.

Thermodynamic parameters	Definition	Formula
η_c	Efficiency of the compressor (%)	$\eta_c = (T_2' - T_1)/(T_2 - T_1)$
η_D	Efficiency of expansion (%)	$\eta_D = (T_3 - T_4)/(T_3' - T_4')$
ρ_{com}	The efficiency loss of the combustion process (%)	$\rho_{Com} = (p_3/p_2)^{(\gamma-1)/\gamma}$
γ	The ratio of specific heat capacity	$\rho_D = (p_6/p_4)^{(\gamma-1)/\gamma}$
ρ_D	The fluidity loss of the turbocharge process (%)	$\rho_D = (p_6/p_4)^{(\gamma-1)/\gamma}$
ρ_R	The fluidity loss of the heat recovery process (%)	$\rho_R = (p_5/p_6)^{(\gamma-1)/\gamma}$
θ_C	The compressor pressure ratio	$\theta_C = T_2'/T_1 = (p_2/p_1)^{(\gamma-1)/\gamma}$
θ_D	The expansion ratio parameter	$\theta_D = T_3/T_4' = (p_3/p_4)^{(\gamma-1)/\gamma} = \theta_C \rho_{Com} \rho_R \rho_D$
α	The temperature ratio parameter	$\alpha = T_3/T_1$
β	The turbocharger parameter	$\beta = (T_4 - T_6)/(T_2 - T_1)$

TABLE 2: The standpoint parameters at points of 1-6 as follows.

State point	P	V	T
1	P_1	v_1	T_1
2	$P_2 = (\theta_C)^{\gamma/(\gamma-1)} P_1$	$v_2 = [1 + (\theta_C - 1)/\eta_C] (\theta_C)^{-\gamma/(\gamma-1)v_1}$	$T_2 = [1 + (\theta_C - 1)/\eta_C] T_1$
3	$P_3 = (\rho_{Com} \theta_C)^{\gamma/(\gamma-1)} P_1$	$v_3 = \alpha (\rho_{Com} \theta_C)^{-\gamma/(\gamma-1)v_1}$	$T_3 = \alpha T_1$
4	$P_4 = P_1 (\rho_R \rho_D)^{-\gamma/(\gamma-1)}$	$v_4 = \alpha (\rho_R \rho_D)^{\gamma/(\gamma-1)} \{1 - \eta_D [1 - (1/\theta_C \rho_{Com} \rho_R \rho_D)]\} v_1$	$T_4 = \alpha \{1 - \eta_D [1 - (1/\theta_C \rho_{Com} \rho_R \rho_D)]\} T_1$
5	$P_5 = P_1$	$v_5 = \alpha_R v_1$	$T_5 = \alpha_R T_1$
6	$P_6 = (\rho_R)^{-\gamma/(\gamma-1)} P_1$	$v_6 = v_1 (\rho_R)^{\gamma/(\gamma-1)} \{ \alpha (1 - \eta_E [1 - 1/\theta_C \rho_{Com} \rho_R \rho_D]) \} T_1$	$T_6 = \alpha \{1 - \eta_D [1 - 1/\theta_C \rho_{Com} \rho_R \rho_D]) - \beta (\theta_C - 1)/\eta\} T_1$

$$\begin{aligned}\varepsilon &= \frac{W + E_T}{E} \\ &= \frac{\eta_{Com}\eta_G[\alpha\eta_D(1 - 1/(\theta_C\rho_{Com}\rho_R)) - \theta_C - 1/\eta_C] + \eta_T\beta(\theta_C - 1)/\eta_C}{\alpha - 1 - (\theta_C - 1/\eta_C)}.\end{aligned}\quad (6)$$

Equation (6) can be written as equation (7).

$$\varepsilon = 1 - \frac{1}{\theta_C} + \beta \frac{\theta_C - 1}{\alpha - \theta_C}.\quad (7)$$

From (7), ε is related to α and θ two parameters, then the partial derivative of ε to θ_C is written as (8) and ε to α is written as (9).

$$\frac{\partial \varepsilon}{\partial \theta_C} = \frac{1}{\theta_C^2} + \frac{\beta(\alpha - 1)}{(\alpha - \theta_C)^2},\quad (8)$$

$$\frac{\partial \varepsilon}{\partial \alpha} = \frac{-\beta(\theta_C - 1)}{(\alpha - \theta_C)^2}.\quad (9)$$

From (8) and (9), there is a nonexistent value for ε , which means ε increases with the increase of θ_C ; ε decreases with the increase of α .

2.3. Mathematical Model for the Turbocharger System with Heat Recovery after Turbocharger. The exergy efficiency ε of the Brayton cycle with heat recovery can be calculated as

$$\varepsilon = \frac{P + W + E_Q}{E}.\quad (10)$$

Equality (10) is very difficult to solve, so the maximum value can only be estimated from the actual work data and error codes, excluding heat treatment.

2.4. Thermodynamic Model for the System with Heat Recovery without Turbocharger. The exergy of heat recovery without the turbocharger is E_Q , which is given as follows:

$$\begin{aligned}E_Q &= \int_{T_5}^{T_4} q_m C_P \left(1 - \frac{T}{T_1}\right) dT \\ &= q_m C_P (T_4 - T_5) - q_m C_P T_1 \ln\left(\frac{T_4}{T_5}\right).\end{aligned}\quad (11)$$

The exergy efficiency ε of Brayton cycle with heat recovery without turbo charger can be calculated as

$$\varepsilon = \frac{P + E_Q}{E}.\quad (12)$$

2.5. Mathematical Process. The mathematical process for the cogeneration system with and without the turbocharger is written as Figure 3. The irreversible influence parameters η_C , η_E , δp_{Com} , η_{Com} , η_G , ρ_{Com} , and ρ_R of the system are defined in the process (gotten as reference). Average field atmospheric pressure and average air temperature and γ are 1.4

and are calculated by the ratio of specific heat capacity at constant pressure C_p (1.004 kJ/kg °C) to specific heat capacity at constant volume C_v (0.718 kJ/kg °C). Table 3 provides the input data required for modeling the system.

Then, α , θ_C , temperature, and pressure of standpoints 2–6 are calculated from Table 1, and the relationship of ε with θ_C and α are calculated through equations (1)–(12) of the irreversible system.

3. Results and Discussion

The effects of ratio and temperature on exergy efficiency with and without turbocharging are investigated to evaluate the feasibility of contract retrofitting in marine engineering systems under various loads. The performance is shown in Figure 4, which shows the performance of the turbodiesel engine in various products after testing the performance of the turbodiesel engine in operation.

3.1. Parameter Tests of the Turbocharger Diesel Engine. CNOOC (National Offshore Oil Corporation of China) received diesel engine certification documents from engine manufacturers after the installation of 111 FPSOs. Table 4 lists the main characteristics of FPSO diesel engines. The approved engine is a marine engineering turbocharged diesel engine, and the acceptance criteria are fully evaluated. The information is certified by BV (Bureau Veritas). According to the information about the permissible limits, the test set-up of the diesel generator is shown in Figure 4. Operating without the variable W, round temperature T1 (K), P1 (bar), working temperature P2 (bar), T4 (K), and T6 (K), etc. are considered. Table 5 shows the operating parameters of the 34025 turbocharger diesel engine.

3.2. The Exergy Efficiency for the Turbocharger System without Heat Recovery. With the irreversible parameters assumed as follows: $\eta_{Com} = 1$, $\eta_G = 1$, $\eta_D = 1$, $\eta_C = 1$, $\rho_{Com} = 1$, $\rho_R = 1$, the relationship of ε with θ_C , α , and β for the turbocharger system without flue gas heat recovery under variable loads is studied as follows.

The relationship between compressor pressure P2 (kg/cm²), fuel consumption mh (kg/h), and air temperature T4 (K) for various nonthermal generators is shown in Figure 5. To confirm the recorded data, T1 (K) and P1 (kg/cm²) are stable for most of the work. As shown in Figure 5, the exergy efficiency increases as the temperature ratio parameter increases, while the pressure ratio is constant. Moreover, irreversible parameters can only be assumed to be constant at 1. For each variable of the load, the irreversibility cannot be less than 1 and the changes are not changed slightly in the

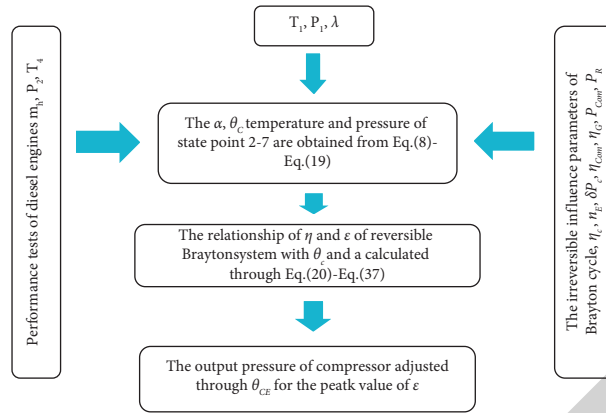


FIGURE 3: The diagram shows the development of the integrated generator in this study.

TABLE 3: Required input data for the modeling of the system.

Parameters	Values	Units
P_1	101.3	Kpa
T_1	298.15	K
γ	1.4	—
C_p	1.004	(kJ/kg °C)
C_v	0.718	(kJ/kg °C)
Q_{net}	42000	(kJ/kg)

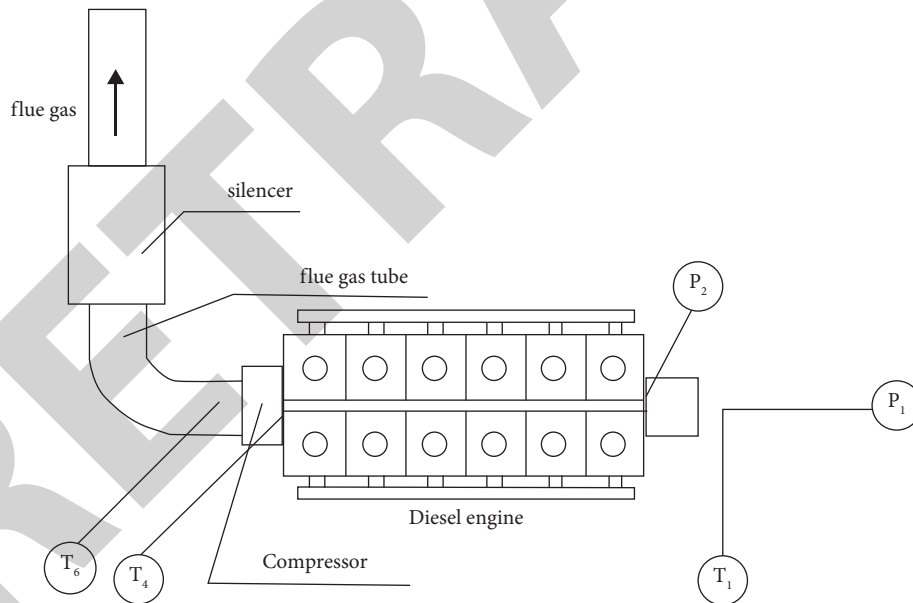


FIGURE 4: The essential operation parameters of turbocharger diesel engines.

TABLE 4: The main parameters of the diesel engine.

Parameters	Values
Model	16CM32C
Speed (rpm)	750
Cylinder bore diameter (mm)	320
Engine type	4-Stroke-cycle-diesel
Power (kW)	8000
Cylinder number	16
Fuel	Oil
Certification	BV

TABLE 5: Operating parameters of 34025 turbocharger diesel engine.

Output power, W (kW)	Load ratio (%)	T_2 (K)	T_3 (K)	q_m (kg/h)	Temperature ratio, α	T_4 (K)	T_6 (K)
8800	110	786.22	1886.71	44098.5	6.06	747.15	619.15
8000	100	761.90	1739.62	45965.8	5.61	708.15	599.15
6800	75	728.04	1603.38	45134.0	5.16	684.15	594.15
4000	50	623.46	1324.7	37760.0	4.28	658.15	620.15
2000	25	507.15	1063.6	30358.3	3.46	644.15	631.15

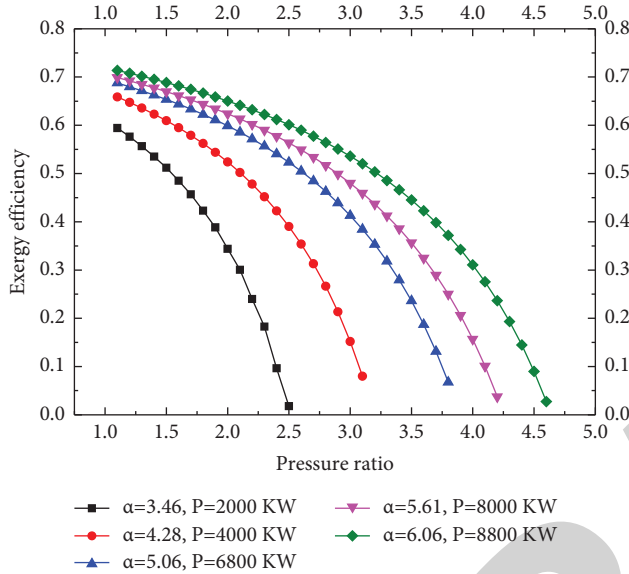


FIGURE 5: The relationship between pressure ratio and exergy efficiency without heat recovery for the turbocharger system.

variable and not influenced by engineering developments. Then, all the irreversible parameters were written on the paper: $\eta_T = 0.5$, $\eta_D = 0.82$, $\eta_C = 0.82$, $\eta_G = 0.97$, $\rho_{Com} = 0.98$, $\rho_R = 1$ (without flue gas heat recovery).

From Figure 6, it can be seen that with the increase of the ratio, the performance of the body first increases and then decreases and the performance of the pressure ratio is higher by 2 to 3.5. Because as the pressure ratio increases, the efficiency of the compressor also increases and the efficiency of the compressor reaches the maximum, and when the pressure ratio increases, the efficiency of the compressor does not increase much, or even increase the oil consumption is increased. The exergy efficiency of the system gradually decreases, so it appears to increase first and then shows the decreasing trend. From Figure 6, compressor P_2 and maximum temperature T_3 of diesel engines can increase the combustion efficiency and the two parameters cannot be unlimitedly increased which can decrease the exergy efficiency ϵ .

3.3. The Exergy Efficiency for the Turbo System with Heat Recovery. The irreversible parameters are shown in Figures 7 and 8.

Figure 7 shows the hot gas reciprocating turbocharger system without consideration of the variable parameters, and Figure 8 shows the hot gas reciprocating the

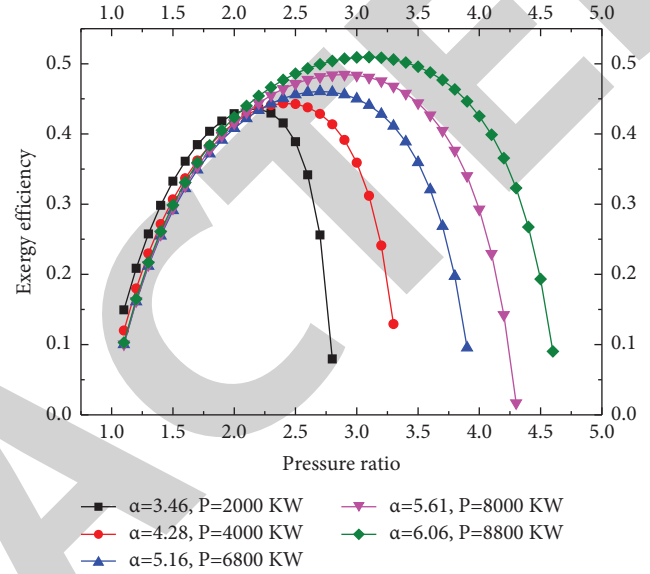


FIGURE 6: The relationship between pressure ratio and exergy efficiency without flue gas heat recovery for the turbo system considering the irreversible parameters.

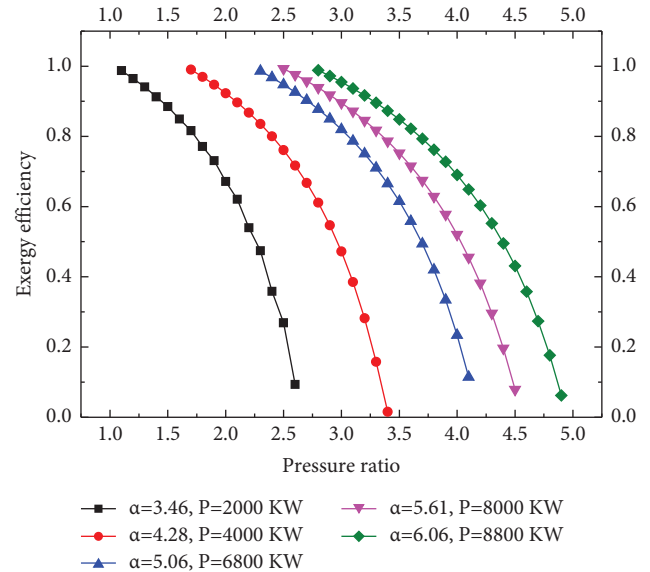


FIGURE 7: The relationship between pressure ratio and exergy efficiency with heat recovery for the turbo system.

turbocharger system with the variable parameters: ($\eta_T = 0.5$, $\eta_D = 0.82$, $\eta_C = 0.82$, $\eta_G = 0.97$, $\rho_{Com} = 0.98$, $\rho_R = 1$). As shown in Figure 7, when the pressure ratio is constant, the

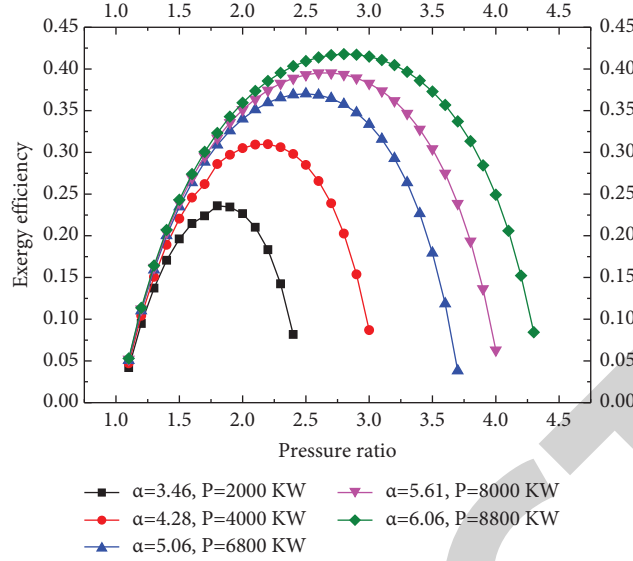


FIGURE 8: The relationship between pressure ratio and exergy efficiency with heat recovery for the turbo system considering the irreversible parameters.

TABLE 6: The θ_{CE} for different systems under variable loads.

Load ratio (%)	α	θ_{CE} (reversible)	θ_{CE} (irreversible)	θ_{CE} (with heat recovery)	θ_{CE} (irreversible, big value)
110	3.46	1.73	1.98	1.4	8.38
100	4.28	2.14	2.31	1.6	12.68
75	5.16	2.58	2.68	1.8	19.44
50	5.61	2.8	2.86	1.9	24.3
25	6.06	3.03	3.03	2.0	30.66

exergy efficiency increases as the temperature ratio parameter increases.

The parameters for the 34025 diesel engine in the real mode are 1.66, 2.02, 2.36, 2.47, and 2.54; for different products, they are 0.4585, 0.5335, 0.5932, 0.6183, and 0.6440, respectively. The peak values of flue gas heat recovery are greater than the peak values of 0.2068, 0.2033, 0.2019, 0.2015, and 0.2049. The reason for this is that the heat recovery can increase the efficiency of the turbocharging system.

The value of the Brayton variable during recovery is 1% lower than the comparison, indicating that high performance can be obtained in real work by adjusting the performance such as pressure ratio and temperature, as shown in Figure 7 and Figure 8.

The different systems under variable loads with heat recovery of the turbocharger system are summarized in Table 6 from Figures 6–8.

Then, (12) can be rewritten as

$$\theta_C = \frac{\eta_D \alpha}{(\alpha \eta_D + 1 - \alpha) \rho_{Com} \rho_R} - \sqrt{\left(\frac{\eta_D \alpha}{(\alpha \eta_D + 1 - \alpha) \rho_{Com} \rho_R} \right)^2 - \frac{\alpha \eta_D (\alpha \eta_C - \eta_C + 1)}{(\alpha \eta_D + 1 - \alpha) \rho_{Com} \rho_R}}. \quad (13)$$

From Figure 8 and Table 4, the thermal efficiency analysis of the marine engineering turbocharger diesel engine is more appropriate than the energy thermal efficiency analysis. Of course, power generation is more important than heat recovery.

3.4. The Comparison of Exergy Efficiency of Heat Recovery with and without the Turbo Charger. Figure 9 shows the calculation results of the waste heat recovery efficiency

without considering the irreversible loss. Figure 9 shows that when the system only heats up again, as the pressure ratio increases, the increase is increased, the pressure is lower, the pressure ratio is higher from 1 to 1.5, and the result is better than the turbocharger. The reason for this is that the turbocharger system converts heat energy into all kinds of energy.

Figure 10 shows that when the system is only reheated, the exergy efficiency first increases and then decreases as the pressure ratio increases, but when the

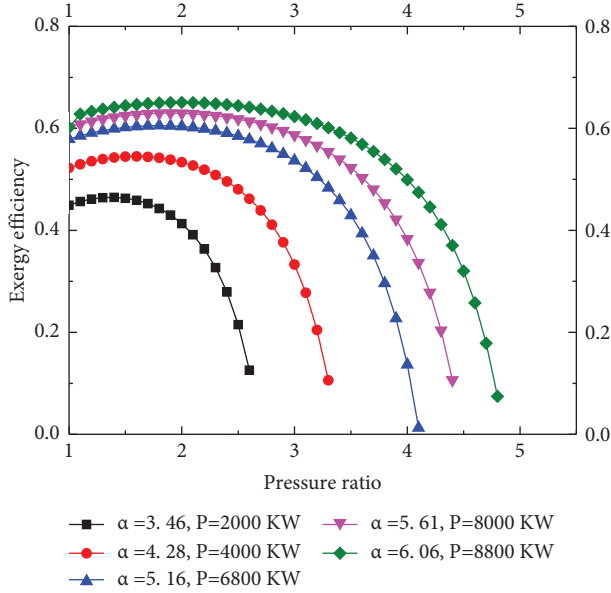


FIGURE 9: The relationship between pressure ratio and exergy efficiency with heat recovery without the turbocharger considering the irreversible parameters.

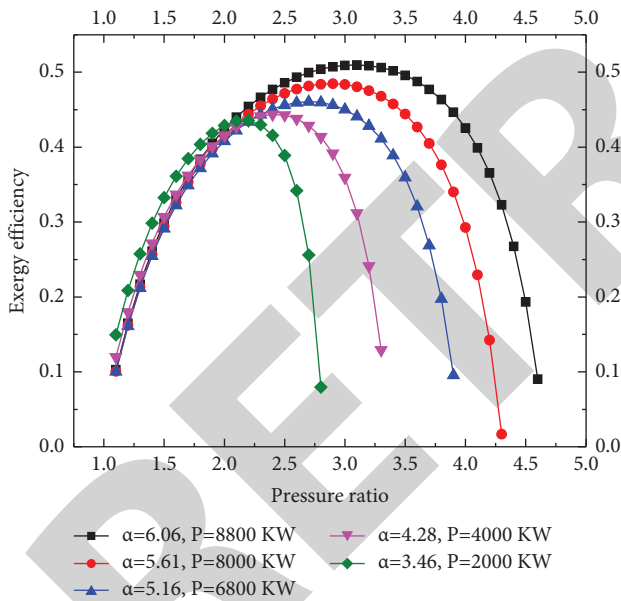


FIGURE 10: The relationship between pressure ratio and exergy efficiency with heat recovery with the turbocharger considering the irreversible parameters.

pressure ratio is moderate, the increase is small and high exergy efficiencies are 0.2 and 3.5, and it is found that when only waste heat recovery is used, its efficiency is higher than that of a compressor system.

4. Conclusions

An analysis of the Brayton system for turbocharged diesel engines has been carried out by considering the effects of local instability, system pressure ratio, and operating

temperature. For the development of marine engineering cogeneration systems and by studying turbochargers under different loads, conclusions can be drawn as follows:

- (1) Compressor P2 and the maximum temperature T3 of the diesel engine can increase the mixing efficiency, and both cannot increase, which reduces the efficiency of exergy.
- (2) It can be seen from the article that by correcting the performance parameters such as pressure ratio and temperature of the turbocharger diesel system, it can achieve high efficiency in operation really.
- (3) If a diesel engine is selected first, cogeneration can be used for heat recovery after the turbocharger for heat and electricity needs. If the use of electricity is chosen first, the use of renewable energy can be used without a turbocharger by installing cogeneration for heat and electricity needs.

Nomenclature

Q_H :	Combustion processes energy (kJ)
Q_v :	Heat recovery quantity (kJ)
Q_L :	Flue gas discharged (kJ)
η_c :	Efficiency of the compressor (kJ)
T :	Temperature (kJ)
η_D :	Efficiency of expansion (%)
ρ_{com} :	The efficiency loss of the combustion process (%)
P :	Pressure (bar)
γ :	The ratio of specific heat capacity
ρ_D :	The fluidity loss of turbo charge process (%)
ρ_R :	The fluidity loss of heat recovery process (%)
θ_C :	The compressor pressure ratio (%)
θ_D :	The expansion ratio parameter (%)
α :	The temperature ratio parameter
β :	The turbo charger parameter
α_R :	Temperature ratio parameter of heat recovery
ε :	The exergy efficiency of Brayton system (%)
W :	The output power of Brayton cycle (kW)
q_m :	Flow mass of flue gas (kg/h)
C_p :	The specific heat at of flue gas
E_T :	The exergy of the turbocharger (kW)
η_T :	The efficiency of turbo (%)
η_G :	The efficiency of the electric generator (%)
E_Q :	Exergy of heat recovery after the turbocharger (%)
E :	The fuel energy (combustion processes) exergy of the Brayton cycle (kW)
θ_{CE} :	The output pressure of compressor (kg/cm ²).

Data Availability

The data used to support the findings of this study are available from the corresponding author upon request.

Conflicts of Interest

The authors declare that they have no conflicts of interest.

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