

Research Article

Joint Effect of the Nonlinearity of Elastic Foundations and the Variation of the Inertia Ratio on Buckling Behavior of Prismatic and Nonprismatic Columns Using a GDQ Method

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In this paper, we present a simple, powerful, yet efficient and easily applicable technique based on the GDQ method for solving nonlinear problems. The proposed technique is implemented to some nonlinear engineering problems in structure analysis. The results reveal that the proposed technique is effective. Then, the proposed technique is used to explain the effects of the variation of cross section area on the nondimensional critical buckling loads for columns with and without elastic foundation for three sets of boundary conditions. Finally, the proposed technique is used to investigate the effect of the nonlinearity term of Winkler elastic foundation on the nondimensional critical buckling loads of nonuniform columns resting on elastic foundations. The effectiveness of the proposed technique is validated through comparing the present results with exact solutions and other numerical results available in references. The proposed method benefits the optimum design of columns against buckling in engineering applications. The most important conclusions from this paper can be summarized as follows. When the inertia ratio varies parabolically, the nondimensional critical buckling loads increase in comparison with varying linearly. Moreover, the nondimensional critical buckling loads increase of the elastic foundation.

1. Introduction

It is pivotal in structural analysis and its design to study the buckling behaviors and to determine the critical buckling loads for uniform and nonuniform structural members.

Many types of structures and structural members for buckling analysis can be defined as a uniform and/or nonuniform with different end conditions widely used as columns in many engineering structure applications such as but not limited to columns, shells and plates, cranes, and other application fields. One of the very important branches of studied in the fields of structural, mechanical engineering, and aeronautical engineering is the buckling analysis of nonuniform columns. So, this field has become more and more systematic during the last decades. This field has become the focal point of many researchers and studies. Several researchers have studied the buckling analysis of nonuniform columns that are closely related to the fields of structural, mechanical, and aeronautical engineering. Until now, a great deal of literature has been published on investigating the buckling of nonuniform columns. This is mainly due to the fact that it may provide an economical solution to carry the desired higher compressive loads in engineering structures.

In 1778, Leonard Euler determined the critical buckling of columns under effect of axial compressive forces in statics [1]. In this field, after Euler study, Swenson [2] studied the nonuniform beams and columns using a simple DC network analyzer, and Chajes [3] presented in his book the principles of structural stability. Greenhill [4] made adept contributions to the field through his study of the exact analytical solution for the buckling problem of a nonuniform column with arbitrary distributions of flexural stiffness and axial distributed forces difficult to establish. A summary of the solutions for some simple cases by several researchers are shown in Coşkun [5].

On the other hand, the closed-form solution of the 4th order differential equation governing for the buckling or vibration behavior of columns or beams with arbitrary distributions of flexural stiffness and different end conditions is very difficult to determine in most of the cases and exists only for limited cases. In addition to this research field, several researchers presented some exact solutions to study the buckling of nonuniform columns, in terms of logarithmic and trigonometric functions, Bessel functions, Lommel functions, and in terms of series representation shown by Eryılmaz et al. [6]. During the last decades, the columns' buckling has become the center point of study for many researchers. In addition, studying the columns' buckling has already become more and more systematic. Later on, analytical approximate techniques were used for the stability analysis of elastic columns and introduced an exact analytical solution for buckling analysis of non-uniform columns presented by Coşkun and Oztürk [7]. Early studies on the stability of finite columns resting on nonlinear elastic foundations with certain prescribed boundary conditions have been presented by Taha and Abohadima [8], Coşkun and Atay [9, 10], Singh and Li [11], and Huang and Li [12] for determining critical loads of nonuniform columns.

The investigations of elastic buckling load and natural frequency of structure members resting on elastic foundations are also one of the complicated and significant problems in many component designs related to soilstructure interaction such as the foundation of buildings, the pipelines embedded in soil, highway pavements, and the like. In modern engineering problems of structural design, structures on elastic foundations have wide applications. In elastic foundation, the elastic media can be defined by an equally spaced elastic supports of equal rigidity. At any cross section of the bar, the reaction of the medium is proportional to the deflection at that section. In this regard, different types of elastic foundation models like Winkler, Pasternak, and Vlasov were presented. In the case of the Winkler foundation model, reaction forces are proportional to the deflection of a beam at each point of contact and the foundation characteristics are modeled by adopting the system of fixed linear springs. The constant of proportionality of these springs is defined by the foundation modulus, also, known as the subgrade modulus. The Winkler type of foundation is extensively used in practice for solving the problems mentioned above. In this model, the elastic foundation by Winkler's theory was inadequate in different problems since this model overlooks the soil cohesion. In order to improve this weakness, any two-parameter elastic foundation models were developed such as the Winkler-Pasternak foundation. In this model, an additional layer is considered in the widely used Winkler model to accomplish the effect of shear interactions between the springs. Closed-form solutions

to some simple problems of beams on elastic foundation proposed by numerous authors are shown in Yankelevsky et al. [13].

An important study for some different types of elastic foundation models as, Winkler, Pasternak, Vlaslov, Filonenko-Borodich presented by Kerr [14] in his paper. Later on, some researchers studied the buckling analysis of beams or columns resting on elastic foundations using different methods such as Eisenberger and Clastornik [15], Malekzadeh and Karami [16], Coşkun and Atay [9, 10], and Coşkun [5, 17]. Some problems studied in structural analysis and free vibration of uniform and nonuniform beams resting on fluid layer using GDQ method are presented by Abumandour et al. [18, 19]. Also, Abumandour et al. [20, 21] used a new hybrid technique combining GDQM and Newton's method to study the free vibration analysis and determine the deflection of nonuniform beams resting on two nonlinear elastic foundations subjected to axial and transverse distributed force.

In engineering applications, the nonlinear elastic foundations provide a simplified model for various complex or complicated nonlinear systems. The nonlinearity sources and the classification were later studied by Kerschen et al. [22] and Afsharfard and Farshidianfar [23]. In his books, Nayfeh [24, 25] presented the weak nonlinear problems. Some researchers used the perturbation techniques to study many cases of the nonlinearity problems of structure analysis. Nowadays, some researchers are interested in using analytical or numerical techniques to solve nonlinear problems. This is mainly due to the fact that every physical problem is really nonlinear problem which means that it should be described using nonlinear equations. However, the analytical solution of nonuniform columns or beams needs the solution of differential equations with variable coefficients, so it is very hard work. There are many numerical and semianalytical methods; however, some analytical methods were developed to obtain approximate solutions for the static and dynamic behavior of non-uniform beams by several researchers and are shown in Coşkun and Öztürk [7] and Adair et al. [26].

The proposed technique presented is based on the GDQM to solve the nonlinear problems. The GDQ method has been applied successfully in our team work for solutions of a variety of problems such as in fluid mechanics [27, 28] and structural analysis [18, 21]. In addition, several researchers are interested in other various problems such as Chen and Cheung [29], Fung [30, 31], Shu et al. [32], and Liu and Wang [33]. A year later, Malekzadeh and Karami [16] exploited a mixed differential quadrature and the finite element method to study the vibration and buckling behaviors of beam-like structures on elastic foundations. Abumandour et al. [20, 21] presented an approach combining the GDQ method with Newton's method to study the free vibration analysis and determine the deflection of nonuniform beams resting on two nonlinear elastic foundations subjected to axial and transverse distributed force.

Mathematical Problems in Engineering

In this paper, we present three objectives. The first objective is to introduce a proposed technique of GDQ method for solving nonlinear problems. The second objective is to apply a proposed technique of GDQ method in determining the nondimensional critical buckling loads of uniform and nonuniform columns resting on the two-layer elastic foundations (linear and nonlinear Winkler elastic foundation and the linear Pasternak elastic foundation). In addition, the paper explains the effects of the varying cross section area on the nondimensional critical buckling loads for columns with and without elastic foundation for three sets of boundary conditions, with different stiffness distributions. Finally, the third objective is to investigate the effect of the nonlinearity term of Winkler elastic foundation on the nondimensional critical buckling loads of nonuniform columns resting on elastic foundations (linear and nonlinear Winkler (normal) foundation and linear Pasternak (shear) foundation), under the three sets of boundary conditions. The validation of the proposed method has been approved by comparing our numerical results with the exact solutions and other available numerical results of uniform and nonuniform columns.

2. Problem Formulations and Mathematical Analysis

Consider a nonuniform column of finite length *L* and width *b* with flexural rigidity *EI*, as shown in Figure 1. The column resting on a Winkler–Pasternak elastic foundation (linear k_1 and nonlinear k_2 Winkler normal stiffnesses and the linear Pasternak shear stiffnesses k_3) continuously restrained along its length subjected to axial load p(x, t) and transverse load F(x, t). The cross section is assumed to vary continuously along the axial direction.

Based on the Euler-Bernoulli beam theory, the governing differential equations of buckling of nonuniform columns resting on nonlinear Winkler-Pasternak elastic foundation are

$$\frac{\partial^2}{\partial x^2} \left(EI \frac{\partial^2 v}{\partial x^2} \right) + \rho A \frac{\partial^2 v}{\partial t^2} + p \frac{\partial^2 v}{\partial x^2} + k_1 v + k_2 v^3$$

$$-k_3 \frac{\partial^2 v}{\partial x^2} = F(x, t), \quad 0 \le x \le L,$$
(1)

where v(x, t) is the flexural deflection, ρ is the density of the beam material, A(x) is the area of the beam section, E(x) is Young's modulus, I(x) is the moment of inertia, p(x, t) is the axial force, k_1 is the linear Winkler foundation parameter, k_2 is the nonlinear Winkler foundation parameter, and k_3 is the linear Pasternak foundation parameter.

General case of column buckling is shown in Figure 2. The geometry, the loading conditions, and the effect of end conditions on the column are shown in Figure 3.

The corresponding boundary conditions along the x = 0 and *L* edges are as follows:

For clamped-clamped supported (C-C):

At $\mathbf{x} = 0$,

$$W(0) = \frac{\mathrm{d}W(0)}{\mathrm{d}x} = 0,$$
 (2)

At $\mathbf{x} = \mathbf{L}$,

$$W(L) = \frac{\mathrm{d}W(L)}{\mathrm{d}x} = 0. \tag{3}$$

For simply-simply supported (S-S):

At $\mathbf{x} = 0$,

$$W(0) = \frac{d^2 W(0)}{dx^2} = 0.$$
 (4)

At $\mathbf{x} = \mathbf{L}$,

$$W(L) = \frac{d^2 W(L)}{dx^2} = 0.$$
 (5)

For clamped-simply supported (C-S):

At $\mathbf{x} = 0$,

$$W(0) = \frac{\mathrm{d}W(0)}{\mathrm{d}x} = 0,$$
 (6)

At $\mathbf{x} = \mathbf{L}$,

$$W(L) = \frac{d^2 W(L)}{dx^2} = 0.$$
 (7)

To obtain the buckling loads, assume the solution of equation (1) to be

$$v(x,t) = V(x)e^{i\omega t},$$
(8)

where the amplitude of free vibration is V(x) and the natural frequency of the column is ω . Substituting equation (8) into equation (1) yields

$$\frac{\mathrm{d}^2}{\mathrm{d}x^2} \left(EI \frac{\mathrm{d}^2 V}{\mathrm{d}x^2} \right) e^{i\omega t} + \rho AV \omega^2 e^{i\omega t} + p \frac{\mathrm{d}^2 V}{\mathrm{d}x^2} e^{i\omega t}$$
$$+ k_1 \left(V e^{i\omega t} \right) + k_2 \left(V^3 e^{3i\omega t} \right) - k_3 \frac{\mathrm{d}^2 V}{\mathrm{d}x^2} e^{i\omega t} = 0, \quad 0 \le x \le L,$$
(9)

$$\frac{\mathrm{d}^2}{\mathrm{d}x^2} \left(EI \frac{\mathrm{d}^2 V}{\mathrm{d}x^2} \right) + \rho AV \omega^2 + p \frac{\mathrm{d}^2 V}{\mathrm{d}x^2} + k_1 V + k_2 \left(V^3 e^{2i\omega t} \right)$$

$$- k_3 \frac{\mathrm{d}^2 V}{\mathrm{d}x^2} = 0, \quad 0 \le x \le L.$$
(10)

For static deflection:

The inertia term is neglected and (10) becomes



FIGURE 1: (a) Geometry of uniform columns resting on two elastic foundations under axial and transverse load. (b) Geometry of nonuniform columns on two elastic foundations under axial and transverse load.



FIGURE 2: General case of column buckling



FIGURE 3: Three end conditions for classical Euler column. (a) For clamped-simply supported column (fixed/pinned ends). (b) For clampedclamped supported column (fixed ends). (c) For simply-simply supported column (pinned ends).

$$\frac{d^2}{dx^2} \left(EI \frac{d^2 V}{dx^2} \right) + p \frac{d^2 V}{dx^2} + k_1 V + k_2 V^3 - k_3 \frac{d^2 V}{dx^2} = 0, \quad 0 \le x \le L.$$
(11)

Equation (11) can be transformed to a nondimensional form as follows:

$$S(X)\frac{d^{4}W}{dX^{4}} + 2\frac{dS(X)}{dX}\frac{d^{3}W}{dX^{3}} + \frac{d^{2}S(X)}{dx^{2}}\frac{d^{2}W}{dX^{2}} + K_{1}W$$

$$+ K_{2}W^{3} - K_{3}\frac{d^{2}W}{dX^{2}} + \lambda\frac{d^{2}W}{dX^{2}} = 0, \quad 0 \le X \le 1,$$
(12)

The nondimensional coefficients are W = V/L, X = x/L, $\lambda = PL^2/EI_0$, $K_1 = k_1L^4/EI_0$, $K_2 = k_2L^6/EI_0$, $K_3 = k_3L^2/EI_0$,

and $S(X) = EI(x)/EI_0$, where W is the nondimensional deflection of the column, P is the nondimensional axial loading, K_1 is the nondimensional linear Winkler (normal) foundation parameter of foundation, K_2 is the nondimensional nonlinear Winkler (normal) foundation parameter of foundation, K_3 is the nondimensional linear

Pasternak (shear) foundation parameter of foundation, the column's flexural rigidity is EI(X), the inertia ratio is S(X), and λ is the nondimensional buckling load of the column.

Equation (12) is a 4th order ordinary differential equation with varying cross section area (inertia ratio). Next, we introduce an overview of the method of solution (generalized differential quadrature method).

3. GDQ Method Review

The DQ method, introduced by Bellman et al. [34, 35], is a numerical technique for solving initial as well as boundary value problems. Bert and his coworkers first used the DQ method to solve problems in structural mechanics in [36]. Bellman et al. [34] suggested two methods to determine the weighting coefficients of the first order derivative. For more details of the DQM and its applications, see [37, 38]. More generally, Shu et al. in [39, 40] presented the generalized differential quadrature (GDQ) and applied it to solve some fluid dynamics problems.

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The dimensionless differential governing equation of column (12) can be transformed into a system of algebraic equations by using the GDQ method [40]. The essence of this method is that it is used to approximate the derivatives of the function W(X) by a weighted linear summation of all the functional values $W(X_i)$ along the computational (discretized) domain. The k^{th} -order derivative of the solution function at grid point *i* in one dimension is expressed by the differential *quadrature rule* as

$$f_{x}^{(k)}(X_{i}) = \frac{\mathrm{d}^{(k)}f}{\mathrm{d}x^{(k)}}\Big|_{x_{i}} = \sum_{j=1}^{N} C_{ij}^{(k)} f(x_{j}), \quad \text{for } i = 1, 2, 3, \dots, N,$$
(13)

where N is the total number of the sampling grid points distribution chosen in the whole domain and $C_{ij}^{(k)}$ are the weighting coefficients for the k^{th} order derivative at the i^{th} sampling point. It should be noted that the weighting coefficients $C_{ij}^{(k)}$ are different at different locations of x_i .

From equation (13), we can realize that the key to GDQ method is how to determine the weighting coefficients for the discretization of a derivative of any order and choose the grid points. It means that the weighting coefficients depend on the order of derivative and on the number of the sampling points (spatial grid spacing) along the domain.

The formula for determining the weighting coefficients can be calculated by using Lagrange interpolation formula as follows [37, 40]:

$$g_k(x) = \frac{M(x)}{(x - x_k)M^{(1)}(x_k)}, \quad \text{where } k = 1, 2, 3, \dots, N,$$
(14)

$$M(x) = (x - x_1)(x - x_2) \cdots (x - x_N),$$
(15)

$$M^{(1)}(x_i) = \prod_{k=1, k \neq i}^{N} (x_i - x_k).$$
(16)

By applying equation (14) at N grid points, they obtained the following algebraic formulations to compute the weighting coefficients A_{ij} . For the first derivative, the weighting coefficients are calculated as follows:

$$A_{ij} = \frac{1}{x_i - x_j} \prod_{k=1, k \neq i, j}^{N} \frac{(x_i - x_k)}{(x_j - x_k)}, \quad j \neq i,$$

$$A_{ii} = \prod_{k=1, k \neq i}^{N} \frac{1}{x_i - x_k}.$$
(17)

After calculating the weighting coefficients of first-order derivatives, the weighting coefficients of higher-order derivatives can be easily determined by simple matrix multiplication which is given as

$$\begin{bmatrix} C^{(k)} \end{bmatrix} = \begin{bmatrix} C^{(1)} \end{bmatrix} \begin{bmatrix} C^{(k-1)} \end{bmatrix} = \begin{bmatrix} C^{(k-1)} \end{bmatrix} \begin{bmatrix} C^{(1)} \end{bmatrix},$$

(18)
 $k = 2, 3, 4, \dots, N-1.$

The accuracy of the results obtained by DQM is affected by choosing the number of grid points, *N*, and the type of sampling points, X_i . It is found that the optimal selection of the sampling points in structure problems, according to Gauss–Chebyshev–Lobatto points, is as follows [41]:

$$X(i) = \frac{1}{2} \left[1 - \cos\left(\frac{i-1}{N-1}\right) \pi \right], \quad i = 1, 2, 3, \dots, N.$$
(19)

4. Discretized Equations and Numerical Implementation

4.1. GDQ Discretized for the Governing Equations of the Column. The GDQ techniques illustrated in the above section 3 can be used to rewrite the differential governing equations into a system of algebraic equations by transforming every space derivative of the dependent variable into the weighted sum of node values as follows:

$$S(X)\left(\sum_{j=1}^{N} D_{ij}W_{j}\right) + 2S^{(1)}(X_{i})\left(\sum_{j=1}^{N} C_{ij}W_{j}\right) + S^{(2)}(X_{i})\left(\sum_{j=1}^{N} B_{ij}W_{j}\right) + K_{1}W_{i} + K_{2}W_{i}^{3} - K_{3}\left(\sum_{j=1}^{N} B_{ij}W_{j}\right) + \lambda\left(\sum_{j=1}^{N} B_{ij}W_{j}\right) = 0, i = 1, 2, 3, ..., N,$$
(20)

where W_i , i = 1, 2, ..., N, is the functional value at the grid X_i , B_{ij} , C_{ij} , and D_{ij} are the weighting coefficient matrix of the second-, third-, and fourth-order derivatives, and $S^{(1)}(X_i)$ and $S^{(2)}(X_i)$ are the second- and first-order derivatives of S(X) at X_i .

4.2. Implementation of Boundary Conditions. Various boundary conditions can be directly implemented by substituting the governing equations using the substitutions of the boundary conditions into the governing equations (SBCGE) method. Then, the problem is simplified to the eigenvalue equation system. The essence of the SBCGE is that the Dirichlet condition is implemented at the boundary points, while the derivative condition is discretized by the GDQ method, a technique referred to as SBCGE [42–44].

The derivatives in the boundary conditions given by equations (2) through (7) can be discretized by the GDQ method. As a result, the numerical boundary conditions can be written as

$$W_1 = 0,$$
 (21)

$$\sum_{k=1}^{N} C_{1k}^{(n0)} W_k = 0, \qquad (22)$$

$$W_N = 0, \tag{23}$$

$$\sum_{k=1}^{N} C_{Nk}^{(n1)} W_k = 0, \qquad (24)$$

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where n0 and n1 may be taken as either 1 or 2. By choosing the value of n0 and n1, equations (21) through (24) can give the following four sets of boundary conditions:

- n0 = 1, n1 = 1-clamped clamped sup ported
- n0 = 1, n1 = 2 clamped simply supported
- n0 = 2, n1 = 1 simply clamped supported

n0 = 2, n1 = 2 - simply - simply supported

Equations (21) and (23) can be easily substituted into the governing equation. We can couple equations (22) and (24) together to give two solutions, W_2 and W_{N-1} , as

$$W_2 = \frac{1}{\text{AXN}} \sum_{k=1}^{N} \text{AXK1} \cdot W_k, \qquad (25)$$

$$W_{N-1} = \frac{1}{\text{AXN}} \sum_{k=1}^{N} \text{AXKN} \cdot W_k,$$
(26)

where $AXN = C_{1,N-1}^{(n0)} C_{N,2}^{(n1)} - C_{1,2}^{(n0)} C_{N,N-1}^{(n1)}$, $AXK1 = C_{1,k}^{(n0)} C_{N,N-1}^{(n1)} - C_{1,N-1}^{(n0)} C_{N,k}^{(n1)}$, and $AXKN = C_{1,2}^{(n0)} C_{N,k}^{(n1)} - C_{1,k}^{(n0)} C_{N,k}^{(n1)}$. Equations (25) and (26) for W_2 and W_{N-1} are expressed in terms of W_3, W_4, \dots, W_{N-2} , and then can be easily substituted into the discrete governing equation (20) which is applied to the interior points $2 \le i \le N - 2$. In order to close the system, the discretized governing equation (20) has to be applied at (N - 4) mesh points. The dimension of the equation system using this approach is $(N - 4) \times (N - 4)$.

$$S(X)\left(\sum_{j=3}^{N-2} D_{ij}W_{j}\right) + 2S^{(1)}(X_{i})\left(\sum_{j=3}^{N-2} C_{ij}W_{j}\right) + S^{(2)}(X_{i})\left(\sum_{j=3}^{N-2} B_{ij}W_{j}\right) + K_{1}W_{i} + K_{2}W_{i}^{3} - K_{3}\left(\sum_{j=3}^{N-2} B_{ij}W_{j}\right) + \lambda\left(\sum_{j=3}^{N-2} B_{ij}W_{j}\right) = 0, i = 3, 4, \dots, N-2.$$
(27)

Equations (27) has (N-4) equations with (N-4) unknowns, which can be written as

$$A]\{W\} + K_2\{W^3\} = \lambda[B]\{W\},$$
 (28)

where, $\{W\} = \{W_3, W_4, \dots, W_{N-2}\}^T$.

5. Numerical Results and Discussion

In this section, the studies have been presented for investigating the buckling behaviors of nonuniform columns resting on two-layer elastic (Winkler (linear and nonlinear) and Pasternak) foundations under axial compression with different boundary conditions using a new hybrid approach of GDQ method. Moreover, the studies have been presented for investigating the buckling behaviors of the nonuniform columns under the effect of the linear (k_1) and nonlinear (k_2) Winkler (normal) foundation parameters and the linear Pasternak (shear) foundation parameter.

The GDQ method is used to compute the critical buckling load with two cases of inertia ratio $S(X) = (1 + \alpha_1 X)^{\alpha_2}$ (the first case: $(\alpha_1 = 1.0)$, $(\alpha_2 = 1.0)$ and the second case: $(\alpha_1 = 1.0)$, $(\alpha_2 = 2.0)$), with three different types of end conditions.

5.1. Validation of the Solution Technique. In order to examine the effectiveness and convergence of the proposed technique, we have determined the critical buckling loads (λ) for a *uniform* and *nonuniform* column subjected to axial compression force, under the three sets of boundary conditions: clamped-clamped (c-c) supported, simply-simply (s-s) supported and clamped-simply (C-S) supported, respectively.

The first validation, for uniform columns, *i.e.*, the inertia ratio S(X) = 1, the numerical results are tabulated in Table 1 together with the exact solution presented by Chajes [3], numerical results using the GDQ presented by Du et al. [45], numerical results using GDQ with SBCGE presented by Ramzy et al. [19], and numerical results using FEM presented by Newberry [46] for various end supports are applied to validate the proposed technique.

By comparing the exact solution and other numerical results with our numerical results of the proposed technique, it is clear that the proposed technique has a rapid convergence and is in very good accordance with the existing numerical results shown in Table 1. In our results, when the number of grid point (N) reaches N = 11, the errors between our numerical results and exact solution are identical to the exact solution up to 15 decimal digits; it means that the proposed technique of GDQ approach is very efficient.

The absolute relative error type in Table 1 represents the accuracy of the proposed technique. This absolute relative error can be defined by the formula $|((Present - Exact)/Exact) \times 100|$.

Figures 4–6 show the first three mode shapes of the uniform column for three types of end supports which results from the proposed technique of the GDQM with SBCGE, respectively.

The second validation for nonuniform columns is to study the effect of the varying cross section area (the inertia ratio (stiffness distributions)), by taking two special cases of the stiffness distributions as polynomials of the form $S(X) = (1 + \alpha_1 X)^{\alpha_2}$: the first case being linearly varying of the inertia ratio $[\alpha_1 = 1.0, \alpha_2 = 1.0]$, i.e., [S(X) = (1 + X)], and the second case being parabolically varying of the inertia ratio $[\alpha_1 = 1.0, \alpha_2 = 2.0]S(X) = (1 + X)^2$. Our numerical results are tabulated in Tables 2 and 3, by using number of grid points (N=11) together with other numerical results presented by Swenson [2], Bleich [47], Bert [48], Du et al. [45], and Ramzy et al. [19], various end supports are applied to validate the proposed technique. Tables 2 and 3 demonstrate that the present critical buckling loads are in very good agreement with the existing numerical results.

			8								
Boundary conditions	Buckling load (λ)										
	Exact solution (Chajes [3])	GDQ (N=11) (Du et al. [45])	FEM (Newberry et al. [46])	SBCGM $(N=11)$ (Abumandour et al. [21])	Present proposed technique $(N=11)$	Absolute relative error %					
Simply-simply	9.8696	9.8696	9.9438	9.8696	9.8696	0.00000					
Clamped- clamped	39.4784	39.4784	39.9730	39.4784	39.4784	0.00000					
Clamped- simply	20.1907	20.19072	20.4972	20.1907	20.1907	0.00000					

TABLE 1: Nondimensional critical buckling loads of uniform columns for various end supports.



FIGURE 4: The first three mode shapes of uniform simply-simply supported column.



FIGURE 5: The first three mode shapes of uniform clamped-clamped supported column.

Other validation for the above case (for nonuniform columns) in order to investigate the effect of the varying of cross section area, by taking the same two special cases of the stiffness distributions as a polynomial of the form $S(X) = (1 + \alpha_1 X)^{\alpha_2}$; the first case being linearly varying of the inertia ratio [S(X) = (1 + X)], and the second case being parabolically varying of the inertia ratio [S(X) = (1 + X)], without elastic foundations, respectively. Our numerical results are tabulated in Tables 4 and 5, by using number of grid points (N = 15) together with other numerical results, by



FIGURE 6: The first three mode shapes of uniform clamped-simply supported column.

an improved version of the Rayleigh technique Bert [48], the stiffness matrix method Eisenberger [49], a low dimensional mathematical model Singh and Li [11], Rayleigh's quotient method Rosa and Franciosi [50], a new and simple approach Huang and Luo [51]. From Tables 4 and 5, it is clear that our results are very compatible with their results. Our results are nearly identical to the results obtained by Huang and Luo [51] and Eisenberger [49].

5.2. Results Using a Proposed Technique of GDQM. In order to investigate the effect of the nonlinearity term of Winkler elastic foundation on the nondimensional critical buckling loads of nonuniform columns resting on twolayer elastic foundations (linear and nonlinear Winkler (normal) foundation and linear Pasternak (shear) foundation), under the three sets of boundary conditions, we study the effect of the varying cross section area (the inertia ratio (stiffness distributions)) by taking two special cases of the stiffness distributions) by taking two special cases of the stiffness distributions as polynomials of the form $S(X) = (1 + \alpha_1 X)^{\alpha_2}$: the first case $[\alpha_1 = 1.0, \alpha_2 = 1.0]$, i.e., [S(X) = (1 + X)], being linearly varying of the inertia ratio, and the second case $[\alpha_1 = 1.0, \alpha_2 = 2.0]$, i.e., $[S(X) = (1 + X)^2]$, being parabolically varying of the inertia ratio, respectively.

Our numerical results are tabulated in Tables 6 and 7, by using number of grid points (N=15) and the nondimensional critical buckling loads (λ) of nonuniform beams

Boundary	Buckling load (λ)									
conditions	Swenson [2]	GDQ (N=11) (Du et al. [48])	SBCGM ($N=11$) (Abumandour et al. [20])	Present proposed technique $(N=11)$						
Simple-simple	14.3	14.511296	14.511255	14.511255						
Clamped- clamped	—	57.3453	57.391536	57.391536						
Clamped-simple	—	29.4406	29.447984	29.447984						

TABLE 2: Nondimensional critical buckling loads of nonuniform columns with linearly varying stiffness distributions S(X) = (1 + X).

TABLE 3: Nondimensional critical buckling loads of nonuniform columns with parabolically varying stiffness distributions $S(X) = (1 + X)^2$.

	Buckling load (λ)									
Boundary conditions	Bleich [47]	Bert [36]	GDQ (N=11) (Du et al. [48])	SBCGM $(N=11)$ (Abumandour et al. [21])	Present proposed technique $(N=11)$					
Simple-simple	20.7923	27.455	20.8047	20.7943	20.7943					
Clamped-clamped	_	—	82.1043	81.9279	81.9279					
Clamped-simple	—	—	41.9679	42.0828	42.0828					

TABLE 4: Nondimensional critical buckling loads of nonuniform columns without elastic foundations with linearly varying stiffness distributions S(X) = (1 + X).

Doundom						
conditions	Bert [36]	Singh and Li [11]	Rosa and Franciosi [50]	Eisenberger [49]	Huang and Luo [51]	Present proposed technique $(N=15)$
Simple-simple	15.31	14.505200922	14.58426	14.511249540	14.5112495395	14.5112495409638
Clamped-clamped	—	57.445246088	—	57.393956136	57.3939561351	57.3939565758955
Clamped-simple	_	29.495964646	_	29.448962806	29.4489628062	29.4489634582706

TABLE 5: Nondimensional critical buckling loads of nonuniform columns without elastic foundations with parabolically varying stiffness distributions $S(X) = (1 + X)^2$.

Poundam						
conditions	Bleich [47]	Singh and Li [11]	Rosa and Franciosi [50]	Eisenberger [49]	Huang and Luo [51]	Present proposed technique $(N = 15)$
Simple-simple	20.7923	20.791633301	21.21653	20.792288456	20.7922884552	20.7922891004959
Clamped- clamped	—	82.228445608	_	81.923363881	81.9233636440	81.9233426420278
Clamped-simple	_	42.313450996	_	42.109176122	42.1091761225	42.1091391192221

TABLE 6: Effects of elastic foundation parameters $[K_1, K_2, K_3]$ on the nondimensional critical buckling loads (λ) for nonuniform columns with linearly varying stiffness distributions S(X) = (1 + X).

	Buckling load (λ)											
	Present proposed technique (N = 15) Elastic foundation parameters			Present proposed technique (N=15) Elastic foundation parameters			Present proposed technique (N=15) Elastic foundation parameters			Present proposed technique $(N=15)$ Elastic foundation parameters		
Boundary conditions												
	$egin{array}{c} K_1 \ 0.0 \end{array}$	$K_2 \\ 0.0$	<i>K</i> ₃ 0.0	K_1 30	K_2 0.0	$K_3 \\ 0.0$	K_1 30	K ₂ 30	$K_3 \\ 0.0$	K_1 30	K ₂ 30	K ₃ 30
Simple-simple Clamped-clamped Clamped-simple	14.5112 57.3940 29.4490			17.5346 59.6607 32.0561			47.5346 89.6607 62.0561			50.1784 92.6179 65.2047		

resting on two-layer elastic foundations, for three sets of boundary conditions. Fifteen nonuniformly spaced grid points were chosen by the previous relation. It can be observed from Tables 6 and 7 that the nondimensional critical buckling loads increase when the column rests on two-layer elastic foundations.

TABLE 7: Effects of elastic foundation parameters $[K_1, K_2, K_3]$ on the nondimensional critical buckling loads (λ) for nonuniform columns with parabolically varying stiffness distributions $S(X) = (1 + X)^2$.

	Buckling load (λ)											
	Present proposed technique (N = 15) Elastic foundation parameters			Present proposed technique (N=15) Elastic foundation parameters			Present proposed technique (N = 15) Elastic foundation parameters			Present proposed technique $(N = 15)$ Elastic foundation parameters		
Boundary conditions												
	K_1	K_2	K_3	K_1	K_2	K_3	K_1	K_2	K_3	K_1	K_2	K_3
	0.0	0.0	0.0	30	0.0	0.0	30	30	0.0	30	30	30
Simple-simple	20.7923			23.7674			53.7674			64.0248		
Clamped-clamped	81.9233			84.1346			114.1346			114.1770		
Clamped-simple		42.1091		44.8215			74.8215			80.4827		



FIGURE 7: The first three mode shapes of nonuniform simplesimple supported column at foundation parameters $(K_1 = 30, K_2 = 30, K_3 = 30)$ with linearly varying stiffness distributions S(X) = (1 + X).



FIGURE 8: The first three mode shapes of nonuniform clampedclamped supported column at foundation parameters $(K_1 = 30, K_2 = 30, K_3 = 30)$ with linearly varying stiffness distributions S(X) = (1 + X).

Form Tables 6 and 7, when the inertia ratio (stiffness distributions) S(X) is varying parabolically, the non-dimensional critical buckling loads of non-uniform columns increase in comparison when the inertia ratio varies linearly.



FIGURE 9: The first three mode shapes of nonuniform clampedsimple supported column at foundation parameters $(K_1 = 30, K_2 = 30, K_3 = 30)$ with linearly varying stiffness distributions S(X) = (1 + X).



FIGURE 10: The first three mode shapes of nonuniform simplesimple supported column at foundation parameters $(K_1 = 30, K_2 = 30, K_3 = 30)$ with parabolically varying stiffness distributions $S(X) = (1 + X)^2$.

The corresponding mode shapes are presented in Figures 7–12. Figures 7–9 show the first case of inertia ratio, and Figures 10–12 show the second case of inertia ratio.



FIGURE 11: The first three mode shapes of nonuniform clamped-clamped supported column at foundation parameters ($K_1 = 30, K_2 = 30, K_3 = 30$) with parabolically varying stiffness distributions $S(X) = (1 + X)^2$.



FIGURE 12: The first three mode shapes of nonuniform clamped-simple supported column at foundation parameters ($K_1 = 30, K_2 = 30, K_3 = 30$) with parabolically varying stiffness distributions $S(X) = (1 + X)^2$.





FIGURE 13: The first three nondimensional buckling loads of nonuniform simple-simple supported columns with various K_2 at $(K_1 = 30, K_3 = 30)$ with linearly varying stiffness distributions S(X) = (1 + X).

FIGURE 14: The first three nondimensional buckling loads of nonuniform clamped-clamped supported columns with various K_2 at $(K_1 = 30, K_3 = 30)$ with linearly varying stiffness distributions S(X) = (1 + X).



FIGURE 15: The first three nondimensional buckling loads of nonuniform clamped-simple supported columns with various K_2 at $(K_1 = 30, K_3 = 30)$ with linearly varying stiffness distributions S(X) = (1 + X).



FIGURE 16: The first three nondimensional buckling loads of nonuniform simple-simple supported columns with various K_2 at $(K_1 = 30, K_3 = 30)$ with parabolically varying stiffness distributions $S(X) = (1 + X)^2$.

To examine the effect of the nonlinear Winkler elastic foundation parameter " K_2 ," we fix the other values of elastic foundations " K_1 " and " K_3 ." Then, we draw " K_2 " versus the nondimensional critical buckling loads of nonuniform columns. It is clear that increasing the nonlinear elastic foundation " K_2 " increases the nondimensional critical buckling loads of nonuniform columns. Figures 8–10 show the first case of inertia ratio, and Figures 11–13 show the second case of inertia ratio.



FIGURE 17: The first three nondimensional buckling loads of nonuniform clamped-clamped supported columns with various K_2 at $(K_1 = 30, K_3 = 30)$ with parabolically varying stiffness distributions $S(X) = (1 + X)^2$.



FIGURE 18: The first three nondimensional buckling loads of nonuniform clamped-simple supported columns with various K_2 at $(K_1 = 30, K_3 = 30)$ with parabolically varying stiffness distributions $S(X) = (1 + X)^2$.

The corresponding Figures 13–18 present the effect of the nonlinear Winkler elastic foundation parameter " K_2 " (see Figures 13–15 for the first case of inertia ratio and Figures 16–18 for the second case of inertia ratio).

In addition, the effect of the nonlinearity term of Winkler elastic foundation (K_2) on the nondimensional critical buckling loads of nonuniform columns for three sets of boundary conditions is studied, and the numerical results are plotted in Figures 19 and 20 with the special two cases of the stiffness distributions $S(X) = (1 + \alpha_1 X)^{\alpha_2}$, for



FIGURE 19: Effects of the nonlinear Winkler elastic foundation parameter " K_2 " on the nondimensional critical buckling loads (λ) for nonuniform columns with linearly varying stiffness distributions S(X) = (1 + X).



FIGURE 20: Effects of the nonlinear Winkler elastic foundation parameter " K_2 " on the nondimensional critical buckling loads (λ) for nonuniform columns with parabolically varying stiffness distributions S(X) = (1 + X).

 $[\alpha_1 = 1.0, \alpha_2 = 1.0]$ and $[\alpha_1 = 1.0, \alpha_2 = 2.0]$, respectively. From Figures 19 and 20, it can be seen that the nonlinear Winkler elastic foundation parameter " K_2 " has a strong influence on the nondimensional critical buckling loads.

6. Conclusion

A proposed approach of GDQ method has been presented to study the buckling behaviors of nonuniform columns resting on the two-layer elastic foundations (linear and nonlinear Winkler elastic foundation and the linear Pasternak elastic foundation). By using the proposed technique of GDQ method, we can determine the nondimensional critical buckling loads for any columns resting on any type of elastic foundations, linear or nonlinear. Several examples of computing the nondimensional critical buckling loads for uniform and nonuniform columns with and without elastic foundation for various boundary configurations have been presented. The validation of the proposed method has been approved by comparing our numerical results with the exact solutions and other available numerical results for uniform and nonuniform columns.

In this paper, we present three endeavors; the first one introduces simpler and efficient approach to simplify the procedure for determining the nondimensional critical buckling loads of nonuniform columns resting on the twolayer elastic foundations (linear and nonlinear Winkler elastic foundation and the linear Pasternak elastic foundation). A proposed approach is presented based on the GDQ method to solve the nonlinear problems. The proposed method is beneficial to optimum design of columns against buckling in engineering applications. The second endeavor explains the effects of the varying cross section area on the nondimensional critical buckling loads for columns with and without elastic foundation for three sets of boundary conditions, by the stiffness distributions as a polynomial of the form $S(X) = (1 + \alpha_1 X)^{\alpha_2}$. Finally, the third endeavor investigates the effect of the nonlinearity term of Winkler elastic foundation on the nondimensional critical buckling loads of nonuniform columns resting on elastic foundations (linear and nonlinear Winkler (normal) foundation and linear Pasternak (shear) foundation), under the three sets of boundary conditions.

This method is simple but efficient for solving similar dynamic problems and free vibration of uniform and nonuniform Euler–Bernoulli beams. By using the proposed technique of GDQ method to determine the nondimensional critical buckling loads for uniform and nonuniform columns with and without elastic foundations for various boundary conditions, the most important conclusions can be summarized as follows:

- (i) When the inertia ratio (stiffness distributions) is varying parabolically, the non-dimensional critical buckling loads of non-uniform columns increase in comparison when the inertia ratio varies linearly.
- (ii) The nondimensional critical buckling loads of nonuniform columns can be increased by supporting the columns by the elastic foundation.
- (iii) The nondimensional critical buckling loads of nonuniform columns can be increased by increasing the nonlinear elastic foundation.
- (iv) From the previous discussion, it can be seen that the nonlinear Winkler elastic foundation parameter has a strong influence on the nondimensional critical buckling loads.

Data Availability

The data used to support the findings of this study are included within the article.

Conflicts of Interest

The authors declare that there are no conflicts of interest with respect to this paper.

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